

AVIATION

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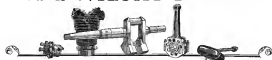
The "Kari-Keen Coupe"

The Curtiss "Challenger"

Parachutes in Civil and Commercial Aviation

AVIATION PUBLISHING CORPORATION
250 WEST 57TH STREET, NEW YORK

There is *NO* Most important part of a WRIGHT ENGINE



"THIS little rocker arm", said a Wright workman, "could cause as big a 'bum' as a broken main rod." The attitude of this man, his realization that his job, no matter how small, involves a responsibility as important as that for the entire engine, is typically Wright. That is what is meant by the expression "Wright engines have no most important part". Every workman throughout the Wright factory knows he is forging one link in a chain, and that the strength of that chain—which perhaps may span an ocean—is dependent upon the link he makes.

Every part—however small—is made with the same care as every other part. The steel for a tiny pin or bolt is heat-treated, machined, inspected, tested and returned with the same scrutiny, the same rigid standards, the same refusal to pass anything but perfection, as that which determines the making of a main connecting rod. This knowledge of responsibility, permeating every individual workman in the

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Engine

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THANK YOU for watching AVIATION



The Oldest American Aeronautical Magazine

Vol. XXV

SEPTEMBER 8, 1932

No. 11

Cooperation Desired

BY the time these few lines appear in print the Aero section branch of the Department of Commerce will have completed the making out of the blank survey questionnaire covering the period of January 1 to June 30 of this year. This questionnaire, while not of great length, requests sufficient information, regarding the activities of the individual operator during the six months period, as to acquire a certain amount of data and case in filling it out. Such being the case, there may be a tendency on the part of some of the operators to procrastinate a little and put off the task until a more "favorable" time.

The Department experienced considerable trouble in securing the filled out questionnaires for the last six months of 1931, and as a result, the publishing of the Department's report covering all survey operations for that period was somewhat delayed. It is to be hoped that there will be a minimum of procrastinating this year and that the Department will receive early returns. Part of the questionnaire is devoted to confidential information, which, of course, will not be made public to regard the individual operator. However, the other part, if properly filled out, will supply information which, when combined with the information supplied by the other questionnaires, will be of considerable interest and value to the entire industry. The completed report as published by the Department will enable one to ascertain the progress of survey operation in this country for the first six months of 1932 and, when compared with similar reports published for 1926 and 1931, to ascertain the progress of survey operations in this country from year to year.

Earning Your Way

IN all of our colleges and universities the majority of students earn all or part of their tuition expenses. Student employment bureaus are well organized, and even in the small towns, where some of the colleges are located, it has been possible for the student to find a part-time job which enables him to earn money and at the same time continue his college course. With the present high cost of obtaining the 300 hr. solo which is required for a transport pilot's license some similar arrangements should be made for the benefit of the flying student who desires making flying his profession.

As present cost flying schools offer to the cash customer, this is, perhaps, all right when all that the student desires is to learn how to solo. However, when he desires to obtain the necessary 300 hr. of flying time in order to qualify for a transport pilot's license the water becomes an entirely different aspect. Two hundred

hours is a lot of time to spend in the air and, because of bad weather, planes being busy, or out of repair, it usually takes a student six months to get to that amount of time. Perhaps that is just as it should be, for flying instruction should be well absorbed, and there is a definite limit to the amount of flying which can be done at one time. However, if schools are going to keep students for the length of time it should be perfectly possible, and also good business, to see that they are provided with part-time jobs so that they can earn as they learn.

As active flying fields there are again a number of jobs to be done which do not require any experience. Many of the students, however, will have had mechanical training, and under supervision could do repair work or fix some position in a nearby factory. If it should be impossible to provide the student with mechanical work the school management would do well to actively assist their students in obtaining part-time jobs outside of the industry. If students are really anxious to learn, and they are willing to work hard and to accept positions which they would ordinarily consider to be beneath their dignity, there is no reason why they should not work their way through a flying course just the same as the student works his way through four years of university training.

The Pilot's Status

ONE often hears the assertion that the pilot's status in the industry will soon be reduced to little more than that of an aerial chauffeur. Such an assertion is rather absurd, for with the growth of air transportation there will be an increase in the size and value of the planes which will have to be piloted. No company is going to entrust a half-million dollar airplane to anyone except an expert of the highest order, and a man of this type will be well paid for his services. The man who pilots the giant airplanes of the future will have to be familiar with radio; he will have to know his engine and its plane; he will have to be an expert on both aerodynamics and navigation, and he will have to have had a vast amount of flying experience.

Most of the time he will not have to use this knowledge, but, like the captain of an ocean liner, he will have to possess the experience and knowledge necessary to meet any emergency that may arise. Coast laws are now secured by satisfactory evidence; they get their bearings by wireless, etc., but the captain has not lost his position. The pilot who is entrusted with a valuable cargo, as a passenger plane, and, above all, the lives of passengers will most definitely have to be as high a type of man as our present air mail pilots, and will undoubtedly receive as high or even higher reward.

The Curtiss "Challenger"

New Six Cylinder Radial Aircraft Engines Developing 170 Hp.
It Designed for the Present OX-5 Powered Planes

By ARTHUR NUTT

Chief Engineer, Curtiss Aeroplane and Motor Co., Inc.

IN the Spring of 1937, it was decided to study the design of a small engine of 170 to 180 hp. This was selected after considering the uses for which the engine would be needed. An engine of this capacity is needed for commercial planes carrying four or five passengers and for training pilots, either in these same planes or in planes designed for that particular purpose. The Curtiss OX-5 engine of 90 hp. weighs approximately 500 lb. It was planned to bring out an engine, which could be used in the same places as the OX-5 engine, with a considerable reduction in weight, and 100 per cent more power.

In considering the design of this engine, low cost and quality were the first considerations. Therefore, an engine with the lowest number of cylinders that would operate smoothly was selected. There were several reasons for selecting six cylinders, rather than an odd number. It was considered that valve gear design was unnecessary, and that a two-stroke more effective pressure would not be possible in the induction system with seven cylinders unless a centrifugal blower was used, which would increase the expense and reduce the reliability of the engine. Five cylinders were not selected for the same reason, and also because it was felt that the engine probably would be too rough with these few cylinders.

In the six cylinder two-stroke crankshaft radial engine, simple induction manifolding is obtained. Three cylinders on each side of the vertical center line are fed from one barrel of a double carburetor. This carburetor is of the simplest form. It has no overhead accelerating device, such as would be necessary if more than three

cylinders were being fed from a single barrel. The fact that the cylinders are equally spaced at 60 deg. angles, provides large spaces between the cylinders for visibility and permits the use of the "Crescent" type of cooling, which is very advantageous as far as heat resistance and proper cylinder cooling are concerned. This is complicated as far as the 1937 English Schneider Cup Racers being an air cooled engine. Another consideration for the selection of the number of cylinders was the simple overhead camshafting. It was possible to manifold three cylinders on each side of the vertical center line in a symmetrical and neat form.

By the use of the two-stroke crankshaft, the crankpin loads are divided, permitting the use of a smaller and shorter crankshaft. With the staggered cylinders the crankshaft can be made smaller in diameter, thereby reducing the over-all diameter of the engine. With a fewer number of cylinders, the bore can be increased and the stroke reduced, still further decreasing the over-all diameter of the engine. A balanced crankshaft is obtained by keeping the counter weights on the extreme crank cheeks opposite with the crankshaft, making a more reliable engine at a reduced cost.

The "Challenger" engine has a bore of 5½ in. and a stroke of 4½ in., giving a cubic displacement of 603 cu. in. Its lowest model designation is R-600. It is a conventional six cylinder radial engine with two valve cylinders, the valves being operated by push rods. The valves are made of special aluminum steel, and act on bronze lapped seats. The cylinders are arranged radially 60 deg. apart, so that actually, it is two three cylinder

engines staggered to get proper firing intervals. The cylinders are of conventional design, and are constructed with cast aluminum heat treated alloy heads with ironed in steel sleeves. The cooling of these cylinders has been worked out very carefully to obtain the maximum amount of weight with maximum cooling.

The rocker boxes are cast integral with the cylinder heads. The rocker arms are mounted in shafted type ball bearings. The covers on the rocker boxes are held securely by means of two flange head cap screws, permitting removal, which gives access to the valve trigger



Front view of the new Curtiss R-600 engine, showing the small crankcase.

adjustment on the end of the rocker arm. The adjusting screw is fitted with a ball with cylindrical integral extension, which bears against the end of the valve stem, reducing the side thrust of the valve stem, and increases the life of the seating points. Push rods are of aluminum with hardened steel ball ends.

The pistons are of the regular Curtiss ribbed design with five piston rings; two compression and two of scraper rings above the piston pin, and one oil scraper ring below it. The pistons have diamond heads to obtain a compression ratio of 14/1 to 1, which also gives a large volume inside of the piston for the collection of oil after the engine is stopped, so that oil cannot run into the combustion chambers through the oil drain holes in the pistons.

Two master connecting rods, each fitted with two short rods, are used. The master rod has four large pins in the cap for attaching the outer section of the rod. There are two sets of lugs on the cap for attaching the short rods. The design makes a very rigid cap with good support for the ball joint, steel ball connecting rod shaft. The bearings for the short rods are bronze bushings. Case hardened wood pins are used. All rods are heat treated steel of "H" section design.

The crankshaft is of the two-piece type with counter weights forged to the two extreme cheeks. The first engines were built with a small counter-weight to decrease the dead weight of the shaft, but production engines will be built without these counter-weights. The shaft is mounted on three ball bearings, the one nearest the propeller being used as a combination radial and thrust bearing. There is no center main bearing.

The crankcase is made in two halves, being split through the center of the front main cylinder, a very rigid construction. The rear half of the crankcase is cast integral with the induction manifolds and accessory drive gear housing. Four bosses for mounting the engine are cast integral with the induction passages of the crankcase. Particular attention was paid in designing the crankcase to make an easy of the hardened surfaces as is possible on the wear place. All induction ports, and the pad for mounting a single aircraft machine gun control drive, which is supplied for military training purposes, are the same distance from the center of the crankcase. This also applies to all cylinder faces.

All accessories are mounted at the rear of the engine on the crankcase, or on the accessory cover. All oil lines are driven through spun gears with the exception of the generator and the single machine gun control drive, which are furnished as extra equipment and are driven through level gears. The design lends itself to production, and gives maximum quality. No adjustments of the spun gears are necessary, which reduces maintenance and assembly costs. Provision is made for driving two three-cylinder, six cylinder Scimitar engines, the present and upcoming all pump, a C-5 gun gasoline pump, a standard generator, a standard starter, one NAAU Stromberg compressor, and the one gun control.

When originally designed, the engine used a single five lobe cam, driven by four spur gears through a jack shaft, thereby doing away with all internal gearing. It was found, however, that the engine would not run smoothly,



Rear quarter view of the "Challenger," showing how the induction manifolds have been cast in as integral part of the crankcase.

or develop as power with the firing order dictated with five lobes. Therefore, two five lobe cams of identical design, locked together, were substituted to give even firing at each manifold. Details of the design of this one mechanism, which runs at one-third the engine speed, are due to A. H. Link, Project Engineer on this engine.

The induction line is attached to the crankcase at the bottom and at the rear, in order that it will be at the lowest point for use with a gravity gasoline system. The elbow connection is exhaust protected. The exhaust heat from

(Continued on page 836)



Front quarter view of a Fairchild cabin monoplane, powered with a new Curtiss "Challenger" engine.

"Liberty Bell" Biplane

New Three Place Plane with Combined Cabin and Cockpit Arrangement Has a High Speed of 135 M. P. H.

A NEW three place biplane surpassing an unusual combination cabin and cockpit arrangement has been developed by the North Star Aircraft Corp. of St. Cloud, Minn. The "Liberty Bell" Type 100 H 4a, as it is designated by the manufacturer, has an enclosed cockpit in the forward part of the fuselage and an open cockpit behind it. The power plant is a Wright Hispano Type B engine developing 182 hp. at 2550 r.p.m. Production of this plane is expected to start in the near future.

The new Liberty Bell was designed by Capt. W. H. Hensen, Minister for general transportation services including passenger, mail and express work. It is of conventional construction having wood wings, welded steel fuselage and light type landing gear. Both upper and lower wings have a span of 35 ft. 10 in. and a chord of 68.5 in. The overall length of the plane is 22 ft. 10 in., and the overall height 9 ft. 6 in. The plane weighs 1335 lb., empty and carries a payload of 450 lb. In performance the Liberty Bell shows a high speed at sea level of 135 m.p.h. and a landing speed of 40 to 45 m.p.h. It will climb fully loaded to 10,000 ft. in 60 sec. The cruising speed at 2/3 throttle is 125 m.p.h. and the fuel consumption at that speed is 9.4 gal. per hr.

There is ample space for two passengers in the aviator cabin which is located in the front section of the fuselage and streamlined toward the nose. The cabin is entered through a door on the left side of the fuselage and has sliding flyable windows in the sides and a fixed window in the rear. It is constructed of steel tubing framed with Balsa wood and covered with fabric. Behind the passen-



Front view of the "Liberty Bell" biplane.

ger's cabin is the pilot's cockpit which is leather upholstered and like the cabin equipped with a safety belt. A semi-circular leather instrument board accommodates Pioneer tachometer, altimeter, compass, oil pressure gauge, thermometer, gasoline gauge and magnetos and lighting switches. The flooring is heavy Balsa, made outside with balsaed filler. A separate compartment for baggage is built into the fuselage behind the pilot's cockpit and is accessible through a door on the side of the fuselage. The cockpit is entered by means of runways on the wing and a step on the side of the fuselage. Forward seats include tool box, fire extinguisher and navigation

lights. The cabin windows and the position of the seat provide excellent visibility ahead and to the sides.

Steel stick control is provided and the extra stick in the cabin is removable. The ailerons are actuated by a bell crank system and the rudder and elevator by push-pull rods. Substituting bearings are used in the control system and aileron interlocks are provided. Throttle, spark and mixture controls for the engine are also installed and control mechanism is visible for inspection.

No wire bracing is employed in the fuselage which is constructed entirely of aluminum pipe members and tubing in the form of a truss. Fushage and stick dangle are



Side view of the new "Liberty Bell"

lained with balsa and covered with Flugskin fabric. The landing gear is gradually streamlined into the main deck. All tail surfaces are constructed of steel tubing and covered with fabric. The stabilizer is adjustable in flight.

A firewall consisting of two thicknesses of aluminum with asbestos between separated the engine section from the remainder of the fuselage. The engine mounting is built of steel tubing with 1 1/2 in. tubular levers and is not detachable. The mounting is built in the form of a Warren truss. A metal framework attached to engine beams and longtruss supports the cowling which is 16 gauge aluminum and is hinged. Ventilation is provided through louvers in the bottom of the cowling. The engine is equipped with a Hamilton Steel propeller with an adjustable pitch, a battery type booster for starting, carved exhaust stacks and a streamlined manifold. An exhaust heater is located in the cabin.

Tail fin tank, constructed of 22 gauge aluminum, are provided, two of which are built into the wings and the other into the fuselage. The fuselage tank is of 35 gal. capacity and is suspended on chrome molybdenum and Ingersoll. Wing tanks have a combined capacity of 34 gal. and feed gasoline to the carburetor through 3/8 in. copper tubing. Fuel from the main tank is forced up to the wing tanks by an engine inertia pump. The flow of gasoline is controlled by valves in the pilot's cockpit.

Landing gear main struts are hinged to the lower fuselage longtruss and shock absorption is effected by a (Continued on page 838)

Parachutes in Civil and Commercial Aviation

By FREDERICK R. NERLY

TWO important expressions of opinion regarding the parachute in civil and commercial aviation were issued recently by officials in the Aeronautics Branch of the Department of Commerce.

One affects parachute jumping, for purposes other than saving lives in emergencies, and the second deals with the lack of a departmental regulation regarding all passengers carried in air transports to wear the seat belt in preservatives. Maj. Clarence M. Young, director of aeronautics, has advised an interpretation of the air commerce regulations and air traffic rules as they apply to parachute jumping. His interpretation restricts, but does not prohibit, exhibition jumps.

In making the interpretation, Major Young places parachute jumping in the category of acrobatic flying, in that it is an intentional maneuver not necessary to any emergency. Further, parachute the jump, as well as dead load drops for training purposes, he places in the same category as that of dropping objects from planes. The pilot in full responsible should such jumps or drops endanger life or injure property.

San Joaquin Over Certain Areas

The regulations, as interpreted by Major Young, prohibit parachute jumping over the congested areas of any city, town or settlement. They prohibit parachute jumping over any open-air assembly of persons. Likewise, they prohibit parachute jumping from an altitude of less than 2,000 ft. over any established civil airway. They also prohibit parachute jumping at any height within 1,000 ft. horizontally of an established airport or landing field. In addition, the regulations require that a jump, where permitted, shall be made so that the parachute opens above 1,500 ft.

The one objection behind these restrictions on parachute jumping is safety for the jumper and safety for the spectators. The regulations are not formulated because the departmental officials think the parachute is unreliable and may not open. Of course, it is held, the professional jumper will not deliberately place his jump in dangerous territory, but he knows, probably better than anyone else, the extent to which a parachute may be controlled.

Recently, Aviation Chief Michael Mac William F. Scott, of the Naval Air Station at Anacostia, D. C., landed on the first except of an apartment house following a 15,000 ft. jump over the City of Washington as part of his program to break the world's record of 28,000 ft. held by Capt. Albert W. Stevens of the Army Air Corps. Scott's jump was planned at first high altitude with the view of landing on Ball's Field first of all, but if not there, then in some of the sparsely settled territory sur-

rounding. As good as Scott is, he floated over the city, and finally landed near the Capitol. He hardly escaped serious injury in coming to rest in such an unusual attitude and place. Scott's case is an exception.

Under the interpretative plan on the regulations by Major Young, the crowds gathering at the various carnivals and fairs throughout the country no longer will



Miss Jane, noted woman parachute jumper, making an exhibition jump.

without parachute jumps made directly overhead or from a point where the wind will carry the jumper as that he can float down to a point in full view, and within a few feet, of the anxious spectators.

While the Department is bent on covering life and limb through its regulation concerning the parachute (Continued on page 838)

The "Kari-Keen Coupe"

*A Two Place Enclosed Cabin Monoplane for Business or Pleasure
Powered with a Veltic 45 Hp. Radial Engine*

WITH the intention of providing the salesman, the business man and the private owner with an economically operated means of air transportation that can be used to advantage in small fields where grade take-offs and slow landings are essential, Kari-Keen Aircraft, Inc., of Sioux City, Ia., is producing a full enclosed wing two-place cabin monoplane that is a neatly upholstered and finished as a modern automobile.

The Kari-Keen coupe is powered with a 45 hp. 3 cylinder without any stress or outside moving parts, it has a clean cut trim appearance. The upper half of the fuselage is finished in dark blue and a cream colored stripe accents the lower half, which is light blue. The wing, horizontal stabilizer and elevator are painted a cream color, while the vertical fin and the rudder are blue. Although the above color combination is standard at present, other finishes will be optional to suit the taste of the individual purchaser.

Four Bolt Hold Wing

The wing is built as one continuous panel and tapers in section as well as in plan from the hards back, being 7 ft. 6 in. at the center of span and narrowing to about 4 ft. 6 in. at 4 ft. 6 in. It is fastened to the fuselage by four simple steel bolts that can be quickly adjusted as they are always visible. The leading and trailing edges are chrome and steel tubing is used to shape the curved ends of the wing and the tail surfaces. An 8 ft. 6 in. 35 wing section is used with the airtight built up box type with square filletings and the ribs of hollowed or designed to withstand extreme loads under severe conditions. The ailerons, which are set in about 2 ft. from the outer edge of the wing at the end of the 8 ft. rib, have the forward part partially enclosed in a channel set

in the wing so there is no gap when they are moved up or down.

The Kari-Keen coupe is powered with a 45 hp. 3 cylinder radial air cooled Veltic engine, which is forward to the engine section of welded steel tubing that is integral with the fuselage. The engine cradle is of aluminum and extends back to a point even with the front part of the door opening, and an inspection door of ample size is provided on either side. The fuel tank is of aluminum and to the rear of the engine is located a 4 gal. oil tank with the filler cap on the top of the cowling, so as to be readily accessible. A 25 gal. gasoline tank forming the leading edge of the center of the wing and extending to the base of the forward spar has a pipe line that runs at each side of the fuselage, near the outer side of the wing, that can be easily reached by the pilot and passenger when within the cabin of the plane. This is an extra safety measure to ease of a forced landing.

Each side of the landing gear, which employs a shock cord arrangement, is independent of the other and is so placed as to make possible a wide track of 6 ft. The landing gear struts are of streamlined tubing, with a short metal tubing for the axle being fitted into that of the streamlined tubing. The 36 in. x 4 in. struts are over-size so as to facilitate a good landing in soft fields. The tail steel, which has a reasonable shoe, is built up of steel tubing (welded fastened to the stern part) and is mounted on a swivel so as to be easily maneuverable in the ground, with a spring holding the steel in a fore and aft position for landing.

The empennage is built of welded steel tubing with the horizontal stabilizer being fixed and the vertical fin adjustable on the ground (short streamlines) wires for bracing run from the upper and lower surfaces of the

(Continued on page 830)



Side view of the new "Kari-Keen Coupe," made by Kari-Keen Aircraft, Inc.

The Goodyear "Puritan"

*Sister Ship of the "Pilgrim" Has an 86,000 Cu. Ft. Capacity and
Carries a Maximum of Four Passengers*

NUMEROUS unconventional improvements which have been worked out in the designing of ground giant airships have been incorporated in the "Puritan," the first dirigible airship built by the Goodyear-Zeppelin Corp. of Akron, Ohio.

The new dirigible is among the smallest in the world built, but slightly larger than the "Pilgrim," its sister ship, built by the Goodyear Tire & Rubber Co., and known as the world's smallest airship. The "Puritan" was built largely as an experimental craft in which various details of design will be put through rigid tests before being used in larger ships. Many of the improvements in the little dirigible will be used in the giant 6,000,000 cu. ft. Zeppelin-type airships which the company hopes to build for the United States government. Four most lighter-than-air experts at the United States and Germany have been working on plans for such ships for the past four years.

Probably the most interesting feature of the "Puritan," from an engineering standpoint, is the framework of its car and keel. This is built of duralumin girders of unspaced design. Openings in the girders are circular or nearly so, a shape which has been found to produce greatest strength, combined with extreme lightness. The girders are exactly like those which are specified for the proposed giant airships.

The passenger tube, built of sheet duralumin over a girder framework, is supported by a local support such like a flat iron. This keel is contained inside the gas

envelope, making the cabin an integral part of the bag.

The "Puritan" obtains its lifting power from 86,000 cu. ft. of helium gas. Two Ryan-Sixteen four-cylinder radial air-cooled engines developing 70 hp. each drive the ship at a cruising speed of 40 m.p.h. and a maximum speed of 55 m.p.h. The engines are mounted on outriggers so as to maintain them from the part of the ship as much as possible, reducing noise and vibration in the cabin. Carrying the maximum of four passengers, the ship has a cruising range of 350 mi. With but two passengers, the range is 550 mi. The overall length is 128 ft., and the greatest width is 37 ft.

A Single Landing Wheel

One of the innovations in design is a single landing wheel projecting below the center of the cabin. This wheel, mounted on a duralumin frame, acts as a support for the ship when it is on the ground, and permits its riding from one place to another. Because of the fact that the airship is lighter than the air it displaces, the cabin sits on the single wheel in landing in almost perfect equilibrium when compared to that on the wheels of an airplane. Another new development is the addition of a vertical rudder on top of the gas bag, at the tail. This provides greater stability. Additional steadiness is provided by the preflexers during in opposite directions.

The "Puritan" was designed for a number of uses. It

(Continued on page 828)



Front quarter view of the new Goodyear "Puritan" on the ground.

Labor Tickets, Their Forms and Uses

By EDWIN R. DOUGLAS
Consulting Engineer

As was stated in a previous article, the forms of labor tickets are multitudinous. In use, they vary from a three by five inch card to 5x13 inch sheets. One ticket may record the time of one man, or of many, on one job, or on a number of jobs. Samples of some of these forms are given in the illustrations.

Many of the items on these tickets have parallel application to the industries where they were used, but some

features might be so used, as each others covers the work of but one man, or group on one job. Forms five, six and eight are especially intended for this purpose, having spaces for entering the scheduled allowances. Number five is interesting because of the very full amount of data given. This was used on a project where many small pieces, often hundreds or thousands all alike, were machined on each order to close finish, checked by very full inspection. A penalty was imposed for spoiled work, and a bonus paid for time saved. The ticket provides, as front and back, for the engineer's detailed report, and for a statement of results to be transferred to production-schedules, etc., and for computing penalties, bonuses, wages, machine-charges and burden.

Ticket numbers five and six are printed in duplicate. They are performed to one apart, with the top copy left carbonized on the back, and one folded for writing. This is a very good plan. While both copies stay together until job is over, the top copy only is given to the work-

things are essential to all labor tickets. These essential things are:

- (a) A clear indication of the work.
- (b) The block-number of the workman.
- (c) The date when work was done.
- (d) The amount of time spent.
- (e) Space for extending the wages.
- (f) In addition, it is frequently desirable to include:
- (g) The quantities desired and actually produced.
- (h) The number of the machine or work-piece.

Sometimes, still other data is called for. Where orders are prepared in advance, to be scheduled through a control board, it is necessary to combine each ticket to the work of one man, or group of men working together, and to one job, or operation. Of the tickets illustrated, forms one, two and three could not be so used, as they cover several jobs on one ticket. Forms

four, five, six and eight are so used, as each others covers the work of but one man, or group on one job. Forms five, six and eight are especially intended for this purpose, having spaces for entering the scheduled allowances. Number five is interesting because of the very full amount of data given. This was used on a project where many small pieces, often hundreds or thousands all alike, were machined on each order to close finish, checked by very full inspection. A penalty was imposed for spoiled work, and a bonus paid for time saved. The ticket provides, as front and back, for the engineer's detailed report, and for a statement of results to be transferred to production-schedules, etc., and for computing penalties, bonuses, wages, machine-charges and burden.

Ticket number eight is interesting by reason of the manner in which it gives information in the time keeping department. No matter which of these forms of single job tickets be used, it is vitally necessary that the total time of each man, as shown by all his job tickets, be checked and proved against his morning-in and evening-out clock records. This is usually done daily, although it is sometimes done only once a week. Unless this proof is made, undetected errors are sure to occur, so that the total amounts charged to costs do not agree with the

totals paid as wages. With any of the tickets from number four to seven, this means that, at each checkup, all tickets in work must be withdrawn and sent to the time-keeping department, being replaced by continuation tickets, newly written, and put back on the job. When checked daily, this involves a very considerable amount of overtime by central board clerks, as the continuation tickets must be written after close of work one day and all be in place before start of work the next day. If the step is working overtime, this considerably complicates the matter. It is to save the clerical work, the checkup is made but once a week, the difficulty of correcting errors becomes much greater.

Forms are Usually Retained

With the form of tickets illustrated as number eight, these defects are greatly reduced. The original ticket may remain at the job until the work is finished, be then days or even weeks, with no continuation tickets to be written. Yet the time keeping department is entered the job-costs and wages will be in balance. The manner of using this ticket is illustrated below, where four consecutive tickets are shown, for jobs on the same machine. These are here used, for reference (a), (b), (c), (d).

Tickets "a" and "b" are shown as having been completed and sent to the office. Ticket "a" has been figured

and posted; ticket "b" has not yet been figured; ticket "c" has been started in work, but is not yet finished. It is at the machine, with a copy of it at the control-board. Ticket "d" has not been started yet. It is at the control-board as "next job."

It is noted that each ticket has space for two lists of order numbers, part numbers and so forth. One list on the left was put on in the planning department when the ticket was originally written, and is the job, for which the starting and stopping times are given on the ticket. The second or right-hand list, as on ticket "b," is entered on by the control board clerk when he stamps the job "finished," it is a memorandum for the cost department of the next job put in work, as shown by ticket "c." This order number is noted by the cost department against the man's number.

Cost Department Keeps Check

From that moment, all has time as charged against that order every day, until the next card "c" bearing the order number comes through "finished." When "c" does, it is there will have appeared on it by the control board clerk, as a "zero," the "next" job, which is taken from ticket "d" when that is given out for work. In this way the cost department is always aware of what order each man is on, and there is no chance of an hour or two slipping in or out through in error in figuring tickets.

The "blanket" tickets, shown one, two, and three, are useful in cases where certain groups of men work, with little change on certain fixed jobs, or on outside repair

and construction jobs where work is not scheduled, where it is not convenient for the cost department to report on and off their jobs, and where the foreman keeps track of what they are doing and makes report of it daily. While sometimes convenient, it is not as accurate method of reporting time, or checking costs. (Continued on page 805)

FOREIGN ACTIVITIES

Aircraft Exhibit Is Held in Italy

*Most Recent Plans and Engine
Developments Shown
—at Milan Salon*

MILAN, ITALY—A number of interesting exhibits and engines were displayed recently at the Milan Aeronautical Salon and although there were fewer exhibits than last year, many of the displays are worthy of note.

The exhibit of S. A. Agosti, Coastal Aeronautics Corporation, which is dedicated to a public school, has an instructional plant reported to be well equipped, consisted of two infrared planes of recent type, a monoplane, C.I. and a biplane, C.I., both models being here designed by Ing. C. M. Calabrese. Both are powered with 100-160 hp Hispano-Suiza M engines. The exhibit of this firm was supplemented by a number of photographs at the Pirelli S. Pietro aviation field and museum, and by graphic descriptions of their plane school services.

Plane Flies Exhibited

The Aeronautica Italiana exhibited its training plane, the Breda 3 bi model, and two other Breda planes of the training type, in which this firm specializes. This firm also exhibited two small variable winging arrangements, the Breda AT with side-by-side and Breda AD with tandem seats.

A collection plate was occupied by the Ivrea Provincial exhibit. The range of the famous Asco power plants were exhibited beginning with the small 200 hp Asco engine with cylinders in line. The Asco 150 power engine of 150 hp

Landing Field Adjoins Hotel

OXFORD, ENGLAND—A hotel known as "The Dog House" adjoining Princes Golf Course here possesses a large field, it is announced, suitable for the landing of light airplanes. Mr. Chander, the proprietor, states that his property is ready to extend special welcome to those flying in the vicinity. In connection with the new hotel service, "The Argonaut," leading British aeronautical publications, report that "those who are Morning, Afternoon, Evening, Wednesday, Evening, or Sportsmen" like this one.

in the latest effort to the products of this firm. The cylinders are in a line of a V, and two lines and are air cooled. Also came the normal Asco 200 engine and in the Sesto Maria constant-speed light, during the Middle West flight, and also for the Panavia recent exhibiting lights. Another interesting model exhibited was the Asco 150 "vanguard" model, especially designed for planes carrying heavy loads with limited speed. The most powerful model exhibited was the Asco 3000 model. This engine is reported to have passed official tests carried out at the presence of aeronautical committee officials, and will shortly undergo also, an official endurance test.

Rapide Six Six Cylinders

The engine, with 12 cylinders in four rows, revolution, develops 900 hp at 1,800 r.p.m., 1,000 hp at 1,600 r.p.m., 1,200 hp at 1,400 r.p.m. and 1,100 hp at 1,200 r.p.m. It has six carburetors. The lubrication is accomplished by pump for the circulation of the oil by pressure, while two pumps of a like type are used for oil recovery.

Mexican Aero Group Opens Aviation School

MEXICO CITY, MEXICO—An aviation school with 10 students, including 30 women, has been opened here sponsored by the Mexican Aeronautical Association. The school of Mexico, centrally located in the city, was chosen for the climate.

The following courses will be given for the school: aviation, meteorology, navigation, radio, mathematics, liquids and air, aerodynamics, regulations, and repairs. Students will be divided into two groups for instruction purposes. The purpose of the school is to train a professional body of civil pilots for commercial aviation.

The school was formally opened by Engineer Ricardo Ruiz, representative of the Comandante in Chief of the Mexican Aeronautical Association. Engineer Juan Guillermo Villalón, president of the Mexican Aeronautical Association, and chief of the Bureau of Civil Aviation, made the opening address.

British Group to Map City of Rio de Janeiro

RIO DE JANEIRO, BRAZIL—An air map of Rio de Janeiro one of the best, made in cooperation with the world but has been developed by the Pan American Union. An experience of aerial photographers has been suggested in London and is a highly detailed air map of the city, and according to the nature of the contract several more will be required for the completion of the task.

Already a party of 15 experienced pilots, photographers and accountants has arrived in the Brazilian capital. Additional suitable assistance will be obtained locally according to Col. E. T. Behrens of the Royal Air Force of England, who is in charge of the work. Colonel Behrens has had long experience in this class of work as numerous British pilots.

Various types of airplanes can be used in the work. Most and Vickers planes have already been secured, it is reported.

Skins for Peruvian Line

LIMA, PERU—Two skins—Derivier six plane monoplane have been shipped to Elmer J. Fournet of this city. Mr. Fournet expects to operate a regular passenger and mail line between Lima and the City of Salem, Northern Peru, flying at a distance of 300 mi.

THE BUYER'S LOG BOOK

Bowser Fueling System

A SYSTEM which permits fueling from the most convenient point on the field and provides fast and accurate gasoline service without ground or fire hazard, is being manufactured by S. P. Bowser & Co., Inc., Fort Wayne, Ind. The compact fueling unit is built into a substantial welded metal box or pit which is so installed that the top of the tank with the various of the field. Heavy water tight covers, fitted with hinged traps give a solid top to the pit, permitting planes to taxi across it if necessary.

The fueling unit consists of a large tank with 30 ft. of flexible gasoline hose, providing a circular fueling

area on the most by means of a hose, leaving the hose free to indicate the velocity and wind conditions. This also makes fueling impossible, as the hose rolls against the tank in a dead wind. After a short experience with this type of cone a pilot is enabled to judge the wind velocity by noting the angle which the cone makes with the mast.

The lightest type of cone is equipped with a small but powerful anemometer placed at the meeting point of the threads so that the cone follows the cone and passes through it. Thus the angle of the cone indicates the wind velocity at night. Sea-lane conditions are indicated by the position of the beam.

These wind indicators are made in three sizes, the largest of which is 12 ft. long, 3 ft. in diameter and has 6 ft. threads. The medium size cone is 8 ft. long, 2 ft. in diameter and has 4 ft. threads and the small type is 4 ft. long, 1 ft. in diameter and has 2 ft. threads. The two larger sizes can be furnished with electric lights for any specified voltage. The regular equipment includes a 12 in. nipple with welded on the top and coupling on the bottom and so that the cone can be quickly removed on a standard 1 in. pipe nut. Unclad, weather resistant fabric and stainless steel hoses are used, connecting to the long life of the cone and renewals can be purchased separately.

G. & O. Radiators

A NUMBER of aircraft manufacturers are using radiators produced by the G. & O. Manufacturing Co. of New Haven, Conn. These products are light and highly efficient units for water cooled installations and have been widely used by the Army and Navy Departments and in the air mail service. It was this type of radiator which was used on the NC planes that first crossed the Atlantic.

Fittings for G. & O. radiators are either of stainless steel or brass tubing or bronze castings. The tanks



Bowser fueling pit with order breaker away to show best refuel, Xacto meter and piping.

area 100 ft. in diameter, a vapor-proof switch for remote control of the unit, and a Bowser Xacto meter on the discharge line. All piping is complete inside the pit which is furnished on a unit ready to install and it is only necessary to connect gasoline line and vent pipe to the outside of the pit box when making the installation. On larger fields the same pumping unit can be made to serve as two additional pits.

Gasoline flow is controlled at the nozzle preventing overfilling at the gasoline tank and a tip is provided to keep dirt and water out of the nozzle. By the use of these units, clean, filtered gasoline is available at any speed at a speed of 10 to 20 gal. per min. The pit can be easily removed and installed in a new location.

Improved Wind Cone

AN IMPROVED type of wind cone which indicates direction, velocity and condition of wind and can be electrically lighted, has been developed by the Heath Airplane Co., at Chicago, Ill. The Non-Pooling Airport Wind Indicator, as it is called by the manufacturer, is the result of 15 years of experimentation.

The outstanding departure from the old type of wind cone is found in the mounting. Instead of being mounted rigidly to the mast, the large hoop is attached to a vertical



One type of G. & O. Radiator

rod and cone are of special bronze composition. These bronze compositions are stronger than the ordinary brass used in the average automobile radiator. They are also tougher and stronger than copper but have the same resistance to corrosion as copper.



A portion of the Ivrea Provincial exhibit at the Milan Salon. The small seven type engine branch the propeller in the new Asco 150 cylinder, 100 hp engine.

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Champion Aviation Spark Plug
brings a new
factor of safety to
aircraft engines



THE Champion Aviation Spark Plug brings to aviation and aviation a new interpretation of dependability—a new feeling of security.

For, after two years of exhaustive research and the most rigorous tests, Champion has given to aerial navigation the first spark plug specifically designed and built for aircraft engines.

Equally unique in conception and in revolutionary design, Champion Aviation Spark Plug embodies entirely new principles.

The dual insulator of exclusive Champion silhouette is so designed that it cannot be broken in such a way as to interfere with engine operation.

The primary "dense" insulator protects the secondary insulator, while a cooling space separates them. Pre-ignition and cutting-out are practically unobtainable with Champion.

Champion's engineering staff developed the Aviation Spark Plug with the aid of Champion's vast experience and unusual resources. They embody all the confidence of material and manufacturing exclusive to all Champion Spark Plugs.

The eager demand from America and Europe for Champion Aviation Spark Plug is based on results. Hundreds of aviators have proven in tests of every extremity that Champion has revolutionized performance, and brought to the aircraft engine an entirely new factor of safety.

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the ship's efficiency would be maintained at a high point. The Pálgos and Perlas are direct descendants of the smaller "Pony Ship" built in 1919 at Alton. The first ship had a gross capacity of but 35,000 cu. ft., and was fitted with hydroplanes. It was the first passenger-carrying ship to appear on regular schedule over a prolonged route in the United States, being used for a long time on passenger service between Goodwin Field at Los Angeles and the Caroline Islands, 10,000 miles. The trip was a delightful one over land and water. In addition, the Pony Ship was used for sporting schools of fish and whales, for motion picture photographic work, and for United States geographic survey work. It was eventually dismantled after having been completely worn out.

A noteworthy fact in connection with the operation of magnet fisheries is their high degree of safety. During the long periods of activity of the Pony Ship and the Pálgos, no accident occurred and no one was injured.

The "Kari-Keen Coupe"

(Continued from page 832)

weather to the fit and the lower part of the windshield respectively. As the place is comparatively long, the tail sections have been so designed as to receive over control due to the leverage obtainable and at the same time they respond readily to the action of the controls.

The fuselage is built up of welded steel tube construction, a Pratt type cross being used. The fuselage windshield has been so designed as to furnish good visibility in all directions and carriage pillars have been used to cut down the obnoxious view so much as possible. The 35 in.



View from the rear of the "Kari-Keen Coupe"

wide door with a 14 1/2 in. x 30 in. glass opening resembles that of a modern automobile in construction, with the look in the metal played outside handle and a crank window for raising the window. The fuselage is mounted to take care of the door opening, which allows angle room for entry or exit. It is noticeable that provisions have been taken to make the cabin as useful for travel as possible for it is used and run tight and there is a possibility of carrying on a construction without a strain when the windows are closed. The interior of the cabin as well as the removable back seat and the spring cushion is heavily upholstered in mohair, making it seem as rich in appearance as the modern closed automobile.

The 38 in. wide seat affords plenty of elbow room for two and three is ample leg room so the pilot and passenger can travel in comfort on the deep cushions during long trips. To the rear of the seat is a baggage compartment, which will accommodate two bags and small packages. The door handles, control levers, window lifts and all interior fittings have been nickel plated so as to be in harmony with the rest of the interior arrangement of the coupe. On the instrument board are the primary switches and a Consolidated panel with the instruments

Airport Boundary Lights

General Electric manufactures boundary lights of both sizes and multiple types for use at airports and intermediate fields.

These fixtures are designed for dependable service and incorporate several important features. Among them are the use of corrosion-resisting metals and provision for draining of condensate from the inside of the globe before it reaches the inner casing.

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If you are planning to go into the airplane business as a dealer and have not found out what the Swallow dealership has to offer, you are overlooking a good bet. Write or write us today and we'll tell you all about it.

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required by the Department of Commerce in regular equipment.

All of the controls, including that for the rudder, elevator and the ailerons are actuated by push rods, no cables are pulled being used and this positive action is assured whenever desired. The dual drive control can be readily disconnected and the pedals which are of the T type with



Front view of the "Kam-Koen" Coupe

a short bar for the foot rest, can be also disconnected and made to lie on the floor and thus be out of the way. Rudder wires are used so when one pedal is depressed the other will be pushed towards the pilot, ready to be used as needed.

The first Kam-Koen coupe was christened during Aviation Week at Sioux City by Miss Edith Mae Frette of Des Moines, Ia. ("Miss Iowa") and a banquet at that time was also held in the Chamber of Commerce in honor of the organizers, which at the time is the state to produce a plane.

From trial flights by Sven Swanson and Cedric Hakken at the Lewis Airport the following performance figures have resulted:

Performance with full load of 830 lb.

Maximum speed 110 m.p.h.
Cruising speed 90 m.p.h.
Landing speed 38 m.p.h.
Service ceiling 20,000 ft.
Absolute ceiling 21,000 ft.
Climb 1,900 f.p.m.

Performance with full load of 1130 lb.

Maximum speed 100 m.p.h.
Cruising speed 85 m.p.h.
Landing speed 45 m.p.h.
Service ceiling 12,000 ft.
Absolute ceiling 16,000 ft.
Climb 600 f.p.m.
Approximate gasoline consumption per hour 5 gal.

The general characteristics of the Kam-Koen coupe are:

Wing span 36 ft.
Chord, center of span 7 ft., 6 in.
Chord, tip of span 4 ft., 6 in.
Wing area, including ailerons 150 sq. ft.
Aileron area (total) 16 sq. ft.
Span of tail 8 ft.
Area of tail 20 sq. ft.
Length overall 22 ft.
Height overall 6 ft., 6 in.
Weight empty 550 lb.
Weight, fully loaded 1,130 lb.
Pay load 480 lb.
Wing loading, full load 75 lb. per sq. ft.
Designed for a load factor of 3.

Engine 38 hp.
Powered by a 5 cylinder, 43 h.p. radial air cooled Velle engine.

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test, permitting oil to flow into the main bearings, if the oil is too thick to pass through the screen under these conditions. The oil feeds into the crankshaft, and the pressure is controlled at the opposite air pressure end by means of a by-pass valve located in the nose casing. This means of control insures positive pressure at every point in the crankshaft. The adjustment is very accessible.

Holes are drilled in the crankpins for forcing the connecting rods. The two link pins in each connecting rod are fed by positive pressure, through proper placed grooves in the connecting rod bearings. Oil is fed to the piston pins and cylinder walls by the oil which is thrown from the connecting rod bearings. The main bearings are



Side view of the six cylinder "Challenger" engine, showing the staggered cylinders

fed also in this same manner. The plain bearings for all the accessories drive gears are fed with oil, through drilled passages, from the rear main bearing. The case mainshaft bearings are fed under pressure from the front end of the crankshaft. The case and case followers are fed by splash.

The rocker arm bearings are fed by means of "Almaria" connections. The ball ends at each end of the rocker arms are fed, through holes drilled in the arms themselves, from this same source of supply. The main oil collector in a pump, which is attached to the bottom of the two crankcases and is built at such an angle that the oil will drain into the sump, even when the plane is in a gliding attitude. The pressure and scavenging system, and the oil drain plug are located in this sump in a very accessible position. A single scavenging pump removes the oil from the engine through a screen, pumping it into the outside oil tank.

The engine is rated at 170 to 180 hp. at 1800 r.p.m. This propeller speed was selected, since it is the maximum that should be used in airplanes which have a top speed of 180 m.p.h. The fuel consumption is normal, and the oil consumption is low, which is a characteristic of all Gates engines.

The actual design work on this engine was started in May, 1937. The first engine was running on the last day of December, 1937. Since that time it has been subjected to rigid endurance tests, having been run approximately

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230 hp., of which 125 hp. were at propeller speeds above 1500 r.p.m., and 40 hp. were at full throttle at 1800 r.p.m. This same engine is also undergoing further endurance tests, and is in excellent condition.

During flight tests of this engine in the Curtiss "Robin" airplane, the smooth operating characteristics were noticeable. The type of induction system gives very remarkable acceleration from the idling speed of 150 to 200 r.p.m. to full throttle which is very desirable. The engine operates smoothly through its entire range. As an example of



Rear view of the Curtiss "Challenger," showing the mounting of the Stromberg carburetor and the two Scintilla magnets.

its endurance qualities, the values in the first engine have not been surpassed since the engine was first started running on December 31, 1927, and all of the values still hold good as when tested.

The Curtiss Challenger engine was designed to give the maximum in quality and the best possible service, at the same time keeping the cost low by proper maintenance with modern equipment. As a result as shown, the engine requires a minimum number of mechanical operations in the process of manufacture. It has as few parts as possible. The engine has passed an official Government 30 hr. endurance test, and will be in quantity production in the fall. It is expected that the engine will sell at a lower price per hp. than the average engine.

The specifications of the engine are as follows:
Model..... R-600
H.p. (Rated at 1800 r.p.m.)..... 170
Type of Engine..... Six-Cylinder Radial
No. of cylinders..... 6
Arrangement of Cylinders..... 2 radial row of 3 each
Bore..... 5 1/2 in.
Stroke..... 5 1/2 in.
Diameter of engine..... 42 in.
Displacement..... 629 cu. in.
Ignition System..... 2 Scintilla Magnets
Carburetor..... Stromberg M-A-U-J
Fuel Consumption Cruising..... 50 lb. per hour 30 hp.
Oil Consumption..... 0.15 lb. per hour 30 hp.
Speed of Propeller..... 1500 r.p.m. (Crankshaft)
Rotation of Propeller..... Clockwise
Weight of engine (empty)..... 420 lb.



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invites your participation in the

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is to be held to precede the

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December 12, 13 and 14, 1928

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Pictured above is the Eclipse Voltage-Regulated Aviation Generator, Engine-Driven Type, 15 Volt, 25 Ampere Capacity, with separate control box. Also produced in 15 Volt, 15 Ampere Capacity.

ECLIPSE Voltage-Regulated Generators, for use in conjunction with Eclipse Electric Starters, landing and running lights, and other accessories, automatically keep the storage battery properly charged. The charging rate is controlled by the voltage regulator, the amount of current delivered to the battery being dependent upon its state of charge and the external connected load. Overcharging is completely avoided, thus insuring long battery life.

The generators may also be operated disconnected from the battery, to take care of external loads up to their rated capacity. They are light in weight and made entirely according to Eclipse standards of accuracy.

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